


EASA	EMERGENCY AIRWORTHINESS DIRECTIVE	
	<p>AD No.: 2010-0274-E</p> <p>Date: 22 December 2010</p> <p>Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.</p>	
<p>This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].</p>		
<p>Type Approval Holder's Name :</p> <p>Aircraft Industries, a.s.</p>		<p>Type/Model designation(s) :</p> <p>L 23 SUPER-BLANÍK sailplanes</p>
<p>TCDS Number: EASA.A.044</p>		
<p>Foreign AD: Not applicable</p>		
<p>Supersedure: None</p>		
ATA 55		Horizontal Stabilizer – Rear Attachment Fitting – Inspection / Replacement
<p>Manufacturer(s): Aircraft Industries, a.s, LET k.p., LET a.s. and LETECKÉ ZÁVODY a.s.</p>		
<p>Applicability: L 23 SUPER-BLANÍK sailplanes, all serial numbers.</p>		
<p>Reason:</p> <p>Cracks were reported on the rear horizontal stabilizer bracket of two L 23 SUPER-BLANÍK sailplanes.</p> <p>This condition, if not corrected, could result in no longer retaining the horizontal stabilizer in place and consequent loss of control of the aeroplane.</p> <p>For the reasons described above, this AD requires immediate inspection of the bracket located at the top of the fin (drawing No. A 730 420 N) and its replacement depending on findings. As a result of the on-going investigation further mandatory terminating action and/or repetitive inspection is likely to follow.</p>		
<p>Effective Date: 24 December 2010</p>		
<p>Required Action(s) and Compliance Time(s):</p> <p>Required as indicated, unless accomplished previously:</p> <p>(1) Before next flight after the effective date of this AD, inspect the rear horizontal stabilizer bracket critical areas (hinge welding areas) for cracks in accordance with the instructions of Aircraft Industries a.s. Mandatory Bulletin (MB) No. L23/053a.</p>		

	<p>(2) If, during the inspection required by paragraph (1) of this AD, a crack is found, before next flight, replace the bracket in accordance with the instruction of Aircraft Industries a. s. Information Bulletin L23/054b.</p> <p>(3) Within 10 days after replacement, send the damaged bracket to Aircraft Industries a.s. and submit the information requested by Aircraft Industries a.s. MB No. L23/053a.</p>
Ref. Publications:	<p>Aircraft Industries a.s. Mandatory Bulletin No. L23/053a dated 14 December 2010.</p> <p>Aircraft Industries a.s. Information Bulletin No. L23/054b dated 20 December 2010.</p> <p>The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.</p>
Remarks :	<ol style="list-style-type: none"> 1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD. 2. The safety assessment has requested not to implement the full consultation process and an immediate publication and notification. 3. Enquiries regarding this AD should be referred to the Airworthiness Directives, Safety Management & Research Section, Certification Directorate, EASA. E-mail: ADs@easa.europa.eu. 4. For any question concerning the technical content of the requirements in this AD, please contact: Aircraft Industries, a.s. - Na záhonech 1177, 686 04 Kunovice, Czech Republic Telephone: +420 572 817 660, Fax: +420 572 816 112 Email: ots@let.cz.