

EASA	AIRWORTHINESS DIRECTIVE	
	<p>AD No.: 2007-0092</p> <p>Date: 10 April 2007</p>	
No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise agreed with the Authority of the State of Registry.		
Type Approval Holder's Name:		Type/Model designation(s):
AIRBUS SAS		A310, A300-600 and A300-600ST aircraft
TCDS Number: DGAC France No 145; EASA A.014		
Foreign AD: Not applicable		
Supersedure: DGAC AD 85-033-067(B)R3 ; DGAC AD 2001-242(B) ; DGAC AD F-2005-174 (EASA approval No. 2005-6369)		
ATA 05	Ageing Systems Maintenance (ASM)	
Manufacturer(s):	AIRBUS (formerly AIRBUS INDUSTRIE)	
Applicability:	<p>A310 aircraft, all certified models, all serial numbers.</p> <p>A300-600 aircraft, all certified models, all serial numbers.</p> <p>A300-600ST aircraft, all certified models, all serial numbers.</p>	
Reason:	<p>The A310 Airworthiness Limitations Section (ALS) Part 4 and A300-600 ALS Part 4 have been issued to introduce the Airworthiness Limitations applicable to A310, A300-600 and A300-600ST ageing aircraft systems in the frame of aircraft extended Limit Of Validity (LOV). The first issue (Revision 00) of these documents introduced Ageing Systems Maintenance (ASM) limitations defined in accordance with a process agreed between AIRBUS and the European Aviation Safety Agency (EASA) by way of the equipment life extension study.</p> <p>The Subpart 4-3(a) of the ALS Part 4 revision 01, related to the Trimmable Horizontal Stabilizer Actuator, refers to maintenance tasks which satisfy the intent of AD 2001-242(B). Therefore AD 2001-242(B) is cancelled.</p> <p>The Subparts 4-2 and 4-3(a) of the A310 ALS Part 4 revision 01, related to the Flap Screw Jack, refer to life limitations and maintenance tasks which satisfy the intent of AD 85-033-067(B)R3. Therefore AD 85-033-067(B)R3 is cancelled.</p> <p>The Subpart 4-3(b) of the A310 ALS Part 4 revision 01, related to the Flap Transmission Shaft, refers to maintenance tasks which satisfy the intent of CN F-2005-174. Therefore AD F-2005-174 is cancelled.</p>	
Effective Date:	24 April 2007	

Compliance:	<p>Within 3 months after the effective date of this AD, it is mandatory to adhere to the requirements of:</p> <ul style="list-style-type: none"> - AIRBUS A310 ALS Part 4 revision 01 for A310 aircraft, - AIRBUS A300-600 ALS Part 4 revision 01 for A300-600 and A300-600ST aircraft. <p>Note: For Operators who previously complied with the requirements of cancelled Airworthiness Directives 85-033-067(B)R3, 2001-242(B) or F-2005-174 in regard to the life limitations/maintenance tasks, the last accomplishment of each limitation/task must be retained as a starting point for the accomplishment of each corresponding limitation/task interval now introduced in the revision 01 of AIRBUS A310 or A300-600 ALS Part 4 documents.</p>
Ref. Publications:	<p>AIRBUS A310 ALS Part 4 revision 01 approved on 21-Dec-2006; AIRBUS A300-600 ALS Part 4 revision 01 approved on 21-Dec-2006, or later approved revisions of these documents.</p>
Remarks:	<ol style="list-style-type: none"> 1. If requested and appropriately substantiated the responsible EASA manager for the related product has the authority to accept Alternative Methods of Compliance (AMOCs) for this AD. 2. This AD was posted on 14 February 2007 as PAD 06-246R1 for consultation until 07 March 2007. No comments were received during the consultation period. 3. Enquiries regarding this AD should be addressed to the AD Focal Point – Certification Directorate, EASA; E-mail: ADs@easa.europa.eu 4. For any question concerning the technical content of the requirements in this AD, please contact AIRBUS SAS – Airworthiness Office – EAZ Fax: +33 5 61 93 37 44.