


<b>EASA</b>	<b>AIRWORTHINESS DIRECTIVE</b>	
	<p><b>AD No.: 2007-0230</b></p> <p><b>Date: 15 August 2007</b></p>	
No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise agreed with the Authority of the State of Registry.		
<b>Type Approval Holder's Name:</b>		<b>Type/Model designations:</b>
AIRBUS SAS		A310 aircraft
TCDS Number: France No 145		
Foreign AD: Not applicable.		
Supersedure: This Airworthiness Directive (AD) supersedes DGAC France AD 2002-578(B)		
<b>ATA 28</b>	<b>Fuel – Electric / Electronic Common Installation Fuel System Route 2S – Modification</b>	
<b>Manufacturer:</b>	AIRBUS (formerly AIRBUS INDUSTRIE).	
<b>Applicability:</b>	AIRBUS A310 aircraft, all certified models, all serial numbers, except those aircraft on which AIRBUS Service Bulletin (SB) A310-28-2148 revision 2 has been embodied in service (AIRBUS modification Nos. 12427 and 12435).	
<b>Reason:</b>	<p>Within the scope of the Fuel System Safety Program (FSSP), analyses of the wire routing shown that route 2S of the fuel electrical circuit in the RH wing must be modified in order to ensure better segregation between fuel quantity indication wires and the 115 VAC wires of route 2S.</p> <p>DGAC France issued Airworthiness Directive (AD) 2002-578(B) to render these improvements mandatory.</p> <p>The manufacturer recently revised SB A310-28-2148 and additional work is required by the revision 2 for aircraft modified by issue 00 or revision 1.</p> <p>This new AD, which supersedes DGAC France AD 2002-578(B), is issued to require operators to accomplish the additional work introduced in SB A310-28-2148 revision 2.</p>	
<b>Effective Date:</b>	29 August 2007	
<b>Compliance:</b>	<p>(1) Unless already accomplished in accordance with the instructions of AIRBUS SB A310-28-2148 original issue or revision 1:</p> <p>Within 4 000 flight hours after 07 December 2002 [Effective date of AD 2002-578(B)], isolate the route 2S of the fuel system in accordance with</p>	

	<p>the instructions of SB A310-28-2148 revision 2;</p> <p>(2) If already accomplished in accordance with the instructions of AIRBUS SB A310-28-2148 original issue or revision 1 :</p> <p>Within 6 000 flight hours or 30 months after the effective date of this AD, whichever occurs first, perform the additional work incorporated in SB A310-28-2148 revision 2.</p>
Ref. Publications:	AIRBUS Service Bulletin A310-28-2148 original issue, revision 1; and A310-28-2148 revision 2 or later approved revisions.
Remarks :	<ol style="list-style-type: none"> <li>1. If requested and appropriately substantiated the responsible EASA manager for the related product has the authority to accept Alternative Methods of Compliance (AMOCs) for this AD.</li> <li>2. This AD was posted for consultation on 17 July 2007 as PAD 07-127 until 14 August 2007. No comments were received during the consultation period.</li> <li>3. Enquiries regarding this AD should be referred to the AD Focal Point - Certification Directorate, EASA. E-mail: <a href="mailto:ADs@easa.europa.eu">ADs@easa.europa.eu</a></li> <li>4. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS SAS – EAW (Airworthiness Office) Telephone:+ 33 5 61 93 36 96, Fax:+ 33 5 61 93 44 51.</li> </ol>