EASA	AIRWORTHINESS DIRECTIVE	
	AD No.: 2007 - 0214 Date: 07 August 2007	
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		iness Directive applies, except in accordance with the erwise agreed with the Authority of the State of Registry.
Type Approval Holder's Name :		Type/Model designation(s) :
AIRBUS		A318, A319, A320 and A321
TCDS Number : EA	SA A.064	
Foreign AD : Not ap	plicable	
Supersedure : DGA	C F-2005-052R1, EASA ap	proval No. 2005-3071
ATA 25	Equipment/Furnishings – 103VU Box – Hinge Pin Stopper – Modification	
Manufacturer(s):	AIRBUS, formerly AIRBUS INDUSTRIE	
Applicability:	AIRBUS models A318-111, A318-112, A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A320-111, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231 and A321-232, all manufactured serial numbers (MSN) except aircraft that have received:	
	AIRBUS mod	dification 36115 in production or
	AIRBUS Ser	vice Bulletin (SB) A320-25-1535 in service
Reason:	Cut wires and wires damaged by heat effect have been found in several 103VU boxes. Those electrical damages are caused by a contact between the horizontal hinge pin and adjacent electrical harness.	
	Investigation revealed that the pre-mod 36115 manufacturing process used to secure the pin by deforming the hinge ends or the hinge pin ends does not prevent the pin translatory motion.	
	To avoid harness damage and subsequent failure on the associated systems, DGAC Airworthiness Directive (AD) F-2005-052R1, dealing with the same subject, mandated the inspection of 103 VU panel hinge pin according to AIRBUS All Operators Telex (AOT) 25A1440.	
		s shown that AOT 25A1440 inspection procedure tive actions were not sufficient to detect and solve

	This AD supersedes DGAC AD F-2005-052R1 and mandates the installation of a hinge pin stopper on the internal door of the 103VU.	
Effective Date:	21 August 2007	
Compliance:	Before March 31 st 2009, install the hinge pin stopper on the internal door of the 103VU in accordance with the instructions given in SB A320-25-1535.	
Ref. Publications:	AIRBUS Service Bulletin (SB) A320 25-1535 original issue, or later approved revisions.	
Remarks :	If requested and appropriately substantiated the responsible EASA manager for the related product has the authority to accept Alternative Methods of Compliance (AMOCs) for this AD.	
	 This AD was posted on 13 June 2007 as PAD 07-094 for consultation until 04 July 2007. The Comment Response Document can be found at http://ad.easa.europa.eu/. 	
	3. Enquiries regarding this AD should be addressed to the AD Focal Point, Certification Directorate, EASA. E-mail ADs@easa.europa.eu .	
	4. For any question concerning the technical content of the requirements in this AD, please contact AIRBUS - Fax 33 5 61 93 44 51.	