

EASA	AIRWORTHINESS DIRECTIVE	
	<p>AD No.: 2007-0173</p> <p>Date: 18 June 2007</p>	
No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise agreed with the Authority of the State of Registry.		
Type Approval Holder's Name :	Type/Model designation(s) :	
AIRBUS SAS	A310, A300-600 aircraft	
TCDS Number: France No. 145		
Foreign AD: Not applicable		
Supersedure: This AD supersedes EASA AD 2007-0149.		
ATA 53	Fuselage – Fasteners at Aft Fuselage Frame 91 (FR91) – Replacement	
Manufacturer(s):	AIRBUS (formerly AIRBUS INDUSTRIE)	
Applicability:	<p>AIRBUS A310 aircraft, models -304, -308, -322, -324 and -325, all serial numbers, except aircraft which have received in service application of AIRBUS Service Bulletin (SB) A310-53-2126 (AIRBUS modification No 13011) and,</p> <p>AIRBUS A300-600 aircraft, models B4-601, B4-603, B4-605R, B4-620, B4-622, B4-622R, C4-620, F4-605R, F4-622R and C4-605R/F all serial numbers, except aircraft which have received application of AIRBUS modification No 13273 in production or application of SB A300-53-6156 in service.</p>	
Reason:	<p>Due to the recalculation of loads for the Multi Role Transporter and Tanker (MRTT) aircraft, it has been found that a structural reinforcement at the aft section of the fuselage (FR87 ÷ FR91) is required for A300-600 aircraft and A310 aircraft with a Trim Tank installed. Airworthiness Directive (AD) 2007-0149 was issued to render mandatory such structural reinforcement in order to meet regulatory certification strength in this area.</p> <p>This new AD cancels and replaces AD 2007-0149 in order to extend the applicability to A300-600 aircraft models A300F4-622R and A300C4-605R/F. AD applicability extension overrides the SB A300-53-6156 original issue effectivity, in which these aircraft models were omitted.</p>	
Effective Date:	02 July 2007	

Compliance:	Within 2,500 flight cycles following 05 June 2007 [effective date of AD 2007-0149], unless already accomplished, reinforce the aft section of the fuselage in accordance with instructions defined in SB A310-53-2126 or SB A300-53-6156.
Ref. Publications:	AIRBUS Service Bulletin SB A310-53-2126 original issue; and SB A300-53-6156 original issue, or later approved revisions of these documents.
Remarks :	<ol style="list-style-type: none"> 1. If requested and appropriately substantiated the responsible EASA manager for the related product has the authority to accept Alternative Methods of Compliance (AMOCs) for this AD. 2. This AD was posted on 29 May 2007 as PAD 07-087 for consultation until 12 June 2007. No comments were received during the consultation period. 3. Enquiries regarding this AD should be referred to the AD Focal Point - Certification Directorate, EASA. E-mail: ADs@easa.europa.eu . 4. For any question concerning the technical content of the requirements in this AD, please contact AIRBUS SAS – EAW Airworthiness Office, Telephone: +33 5 61 93 36 96, Fax: +33 5 61 93 44 51.