



ÚŘAD PRO CIVILNÍ LETECTVÍ  
ČESKÁ REPUBLIKA  
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## PŘÍKAZ K ZACHOVÁNÍ LETOVÉ ZPŮSOBILOSTI

**Číslo: 2007-0155**

Datum účinnosti: 12. června 2007

**Toto AD nahrazuje rakouské AD  
A-2004-004R2**

**BRP-Rotax GmbH & Co. KG**

Rotax 912 A, 912 F, 912 S, 914 F

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Tento PZZ je vydáván pro výrobek transferovaný pod působnost EASA


Na základě rozhodnutí EASA je následující Příkaz k zachování letové způsobilosti závazný pro všechny výrobky provozované v EU, na které se daný PZZ vztahuje.

Provedení PZZ, který se vztahuje podle typu a výrobního čísla na výrobek je pro provozovatele/vlastníka letadla zapsaného do leteckého rejstříku závazné. Neprovedením PZZ ve stanoveném termínu dojde ke ztrátě letové způsobilosti výrobku.

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*Poznámky:*

- Provedení tohoto PZZ musí být zapsáno do provozní dokumentace letadla.
- Případné dotazy týkající se tohoto PZZ adresujte na ÚCL sekce technická.
- Pokud to vyžaduje povaha tohoto PZZ, musí být zpracován do příslušné části dokumentace pro obsluhu, údržbu a opravy letadla.

<b>EASA</b>	<b>AIRWORTHINESS DIRECTIVE</b>	
	<p><b>AD No : 2007- 0155</b></p> <p><b>Date: 29 May 2007</b></p>	
<p>No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise agreed with the Authority of the State of Registry.</p>		
<p><b>Type Approval Holder's Name :</b> BRP-Rotax GmbH &amp; Co. KG</p>	<p><b>Type/Model designation(s) :</b> Rotax 912 A series Rotax 912 F series Rotax 912 S series Rotax 914 F series</p>	
<p>TCDS Number: EASA.E.121 and EASA.E.122</p>		
<p>Foreign AD: N/A</p>		
<p>Supersedure: Austrian AD A-2004-004R2, EASA Approval Number 2005-6413</p>		
<b>ATA 73</b>	<b>Engine Fuel and Control – Coolant Specification - Modification</b>	
<p>Manufacturer(s):</p>	<p>BRP-Rotax GmbH &amp; Co. KG; Bombardier-Rotax GmbH &amp; Co. KG; Bombardier-Rotax GmbH;</p>	
<p>Applicability:</p>	<p>All versions of the engine type Rotax 912 A, 912 F, 912 S and 914 F</p> <p>These engines are known to be installed on, but not limited to, the following aircraft types:</p> <p>3-i Sky Arrow 650 TC, 650 TCN, 650 TCNS and 710 RG; Aeromot AMT-200 Super Ximango and AMT-300 Turbo Super Ximango; Aircraft Philipp (formerly Alpa-Werke; Nitsche) AVO 68 series Samburo; Aquila AT01; Cessna 150 and A150 series; Diamond (formerly HOAC) H 36 Dimona, HK 36 series Super Dimona, DV 20 Katana and DA20-A1 Katana; Evektor-Aerotechnik EV-97 VLA; Grob G 109; Issoire APM-20 Lionceau; Reims Aviation F150 and FA150 series; Scheibe SF 36R and SF 25C; Stemme S10-VT; Tecnam P 92-J, P 92-JS and P2002-JF; W.D. Aircraft D4 Fascination</p>	

Reason:	<p>Under certain powerplant installation and operating conditions, boiling of conventional coolant with a mixing ratio of 50% coolant and 50% water can occur before reaching maximum permissible cylinder head temperatures (CHT). This can lead to evaporation of the coolant and in consequences to loss of coolant in the coolant system, causing the engine to overheat.</p> <p>This condition, if not corrected could result in engine damage or an accident.</p> <p>Technical investigation shows the possibility to use the conventional 50% coolant and 50% water mixture on specified installations, where due to the installation conditions (radiator installation, radiator size e.g.) on the airframe confirms that the upper limit of 120°C for the coolant (50% coolant- 50% water) will not be exceeded and an evaporation will not occur in the specified limits of operation, these limits –efficiency of coolant system- on the airframe must be demonstrated by the airframe manufacturer due their certification process.</p> <p>This Airworthiness Directive is issued to extend the compliance time on the use of conventional glycol/water coolant in order to allow time for the airframe manufacturer to show compliance for their installation of the coolant system to proof and release the proper coolant.</p> <p>Limitation and restriction for use of glycol/water coolant to max. 120°C and the use of 18 psi (1,2 bar) pressure cap P/N: 922070 before determination of the achievable maximum coolant temperature and cylinder head temperature remain unchanged.</p>
Effective Date:	12 June 2007
Compliance:	<p>To insure safe operation, corrective actions have to be performed on aircrafts with affected engines installed until December 31, 2007:</p> <ul style="list-style-type: none"> <li>- Change of coolant specification: incorporate the mandatory use of waterless coolant into the relevant documentation of the aircraft.</li> </ul> <p>Alternatively the use of conventional coolant is possible. In such case the new operating limit (coolant temperature) has to be applied. The work/compliance has to be performed in accordance with the accomplishment instructions of BRP Rotax Service Bulletin SB-912-043 R2 / SB-914-029 R2:</p> <ul style="list-style-type: none"> <li>- Replacement of the radiator cap</li> <li>- Check cooling system - Efficiency of the cooling system</li> <li>- Determination of the achievable maximum coolant temperature and cylinder head temperature</li> </ul> <p>Effects of these measures on the powerplant installation and on compliance with aircraft related requirements, have to be reviewed by the affected aircraft manufacturers in accordance with aircraft related certification requirements, before these measures are being introduced.</p>
Ref. Publications:	BRP Rotax Service Bulletin SB-912-043 R2 and SB-914-029 R2, dated 10 November 2006, or later approved revision

Remarks :	<ol style="list-style-type: none"><li>1. If requested and appropriately substantiated the responsible EASA manager for the related product has the authority to accept Alternative Methods of Compliance (AMOC) for this AD.</li><li>2. This AD was posted as PAD 07-068 on 30 April 2007 for consultation until 28 May 2007. No comments were received during this period.</li><li>3. Enquiries regarding this Airworthiness Directive should be referred to the AD Focal Point - Certification Directorate, EASA. E-mail: <a href="mailto:ADs@easa.europa.eu">ADs@easa.europa.eu</a> .</li><li>4. For any question concerning the technical content of the requirements in this AD, please contact BRP-Rotax GmbH &amp; Co.KG Ph.: +43 7246 601 0; Fax: +43 7246 601 760 email: <a href="mailto:airworthiness@brp.com">airworthiness@brp.com</a></li></ol>
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