


EASA	AIRWORTHINESS DIRECTIVE	
	<p>AD No.: 2006-0284 R1</p> <p>Date: 13 February 2007</p>	
No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise agreed with the Authority of the State of Registry.		
Type Approval Holder's Name:		Type/Model designations:
AIRBUS SAS		A310 and A300-600 aircraft
TCDS Number: France No 145		
Foreign AD: Not applicable		
Revision/Supersedure: This AD revises EASA AD 2006-0284.		
ATA 24	Electrical Power – Prevention against Fuel Tank Explosion Risks – Fuel Pump Wiring – Modification	
Manufacturer:	AIRBUS (formerly AIRBUS INDUSTRIE)	
Applicability:	AIRBUS A310 and A300-600 aircraft, all certified models, all serial numbers, except for aircraft which have received in production AIRBUS modification 13118 or on which AIRBUS Service Bulletin (SB) A310-24-2097 or A300-24-6094 has been incorporated in service.	
Reason:	<p>Further to the accident of a Boeing 747-131 aircraft (flight TWA800), the FAA has published SFAR 88 (Special Federal Aviation Regulation 88). In their letters referenced 04/00/02/07/01-L296, dated March 4th, 2002 and 04/00/02/07/03-L024, dated February 3rd, 2003, the JAA recommended the application of a similar regulation to the National Aviation Authorities (NAA).</p> <p>Under this regulation, all holders of type certificates for passenger transport aircraft with either a passenger capacity of 30 or more, or a payload capacity of 7,500 pounds (3402 kg) or more, which have received their certification since January 1st, 1958, are required to conduct a design review against explosion risks. This Airworthiness Directive (AD), which renders mandatory the modification of the fuel pump wiring against short circuit, is a consequence of this design review.</p> <p>Revision 1 of this AD is issued to remove AIRBUS A300 aircraft from the applicability. A separate AD, applicable to AIRBUS A300 aircraft, will be issued to take into account the introduction of additional work in AIRBUS SB A300-24-0103 at revision 1.</p>	

Effective Date:	28 September 2006
Compliance:	Before the 31 October 2009, unless already accomplished, modify the inner and outer fuel pumps wiring, route 1P and 2 P harnesses in the LH wing and in the RH wing in accordance with the instructions of Airbus SBs A310-24-2097 or A300-24-6094, as applicable.
Ref. Publications:	AIRBUS Service Bulletins A310-24-2097 original issue or revision 1; or A300-24-6094 original issue or revision 1, as applicable to type; or later approved revisions of these documents.
Remarks :	<ol style="list-style-type: none"> 1. If requested and appropriately substantiated the responsible EASA manager for the related product has the authority to accept Alternative Methods of Compliance (AMOCs) for this AD. 2. Required actions and the risk allowance have granted publication and notification of an immediate AD, ruling out the public consultation process. The original issue of this AD was posted on 26 July 2006 as PAD 06-197 for consultation until 31 August 2006. No comments were raised during the consultation period. 3. Enquiries regarding this Airworthiness Directive should be referred to the AD Focal Point - Certification Directorate, EASA. E-mail: ADs@easa.europa.eu 4. For any question concerning the technical content of the requirements in this AD, please contact AIRBUS SAS – EAW (Airworthiness Office, Ph: + 33 5 61 93 36 96, Fax + 33 5 61 93 44 51).