



ÚŘAD PRO CIVILNÍ LETECTVÍ
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PŘÍKAZ K ZACHOVÁNÍ LETOVÉ ZPŮSOBILOSTI

Číslo: 2006-0203

Datum účinnosti: 19. července 2006

AIRBUS

Modely A318, A319, A320, A321


Tento PZZ je vydáván pro výrobek transferovaný pod působnost EASA

Na základě rozhodnutí EASA je následující Příkaz k zachování letové způsobilosti závazný pro všechny výrobky provozované v EU na které se daný PZZ vztahuje.

Provedení PZZ, který se vztahuje podle typu a výrobního čísla na výrobek je pro provozovatele/vlastníka letadla zapsaného do leteckého rejstříku závazné. Neprovedením PZZ ve stanoveném termínu dojde ke ztrátě letové způsobilosti výrobku.

Poznámky:

- Provedení tohoto PZZ musí být zapsáno do provozní dokumentace letadla.
- Případné dotazy týkající se tohoto PZZ adresujte na ÚCL sekce technická.
- Pokud to vyžaduje povaha tohoto PZZ, musí být zpracován do příslušné části dokumentace pro obsluhu, údržbu a opravy letadla.

EASA	AIRWORTHINESS DIRECTIVE
	<p style="text-align: center;">AD No : 2006 - 0203</p> <p style="text-align: center;">Date: 11 July 2006</p>
No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise agreed with the Authority of the State of Registry.	
Type Approval Holder's Name : AIRBUS	Type/Model designation(s) : A318/A319/A320/A321 Aircraft
TCDS Number : EASA A.064	
Foreign AD : none	
Supersedure : none	
ATA 28	Fuel - Fuel Tank Safety ALS Part 5 - Fuel Airworthiness Limitations (FAL)
Manufacturer(s):	AIRBUS (formerly AIRBUS INDUSTRIES)
Applicability:	<p>AIRBUS aircraft models:</p> <p>A318-111 A319-111 A320-111 A321-111 A318-112 A319-112 A320-211 A321-112 A318-121 A319-113 A320-212 A321-131 A318-122 A319-114 A320-214 A321-211 A319-115 A320-231 A321-212 A 319-131 A320-232 A321-213 A319-132 A320-233 A321-231 A319-133 A321-232,</p> <p>all serial numbers.</p>
Reason:	Subsequent to accidents involving Fuel Tank System explosions in flight (Boeing 747-131 flight TWA800) and on ground, the FAA published Special Federal Aviation Regulation 88 (SFAR88) in June 2001. SFAR 88 required a safety review of the aircraft Fuel Tank System to determine that the design meets the requirements of FAR § 25.901 and § 25.981(a) and (b).

	<p>A similar regulation has been recommended by the JAA to the European National Aviation Authorities in JAA letter 04/00/02/07/03-L024 of 3 February 2003. The review was requested to be mandated by NAA's using JAR § 25.901(c), § 25.1309.</p> <p>In August 2005 EASA published a policy statement on the process for developing instructions for maintenance and inspection of Fuel Tank System ignition source prevention (EASA D 2005/CPRO, www.easa.eu.int/home/cert_policy_statements_en.html) that also included the EASA expectations with regard to compliance times of the corrective actions on the unsafe and the not unsafe part of the harmonised design review results. On a global scale the TC holders committed themselves to the EASA published compliance dates (see EASA policy statement). The EASA policy statement has been revised in March 2006: the date of 31-12-2005 for the unsafe related actions has now been set at 01-07-2006.</p> <p>Fuel Airworthiness Limitations are items arising from a systems safety analysis that have been shown to have failure mode(s) associated with an 'unsafe condition' as defined in FAA's memo 2003-112-15 'SFAR 88 – Mandatory Action Decision Criteria'. These are identified in Failure Conditions for which an unacceptable probability of ignition risk could exist if specific tasks and/or practices are not performed in accordance with the manufacturers' requirements.</p> <p>This EASA Airworthiness Directive mandates the Fuel Airworthiness Limitations, comprising maintenance/inspections tasks and Critical Design Configuration Control Limitations (CDCCL), for the type of aircraft that resulted from the design reviews and the JAA recommendation and EASA policy statement mentioned above.</p>
Effective Date:	19 July 2006
Compliance:	<p>Unless already accomplished, the following actions are rendered mandatory:</p> <p>1. Maintenance/Inspection Tasks:</p> <ul style="list-style-type: none"> - Within 3 months from the effective date of this AD, it is mandatory to strictly adhere to the requirements of <i>AIRBUS</i> ALS part 5, Fuel Airworthiness Limitations Section 1 as defined in document A318/A319/A320/A321 Fuel Airworthiness Limitations, 95A.1931/05 at Issue 1 or later approved version. <p>At the effective date of this AD, for the aircraft in service, defined intervals for FAL have to be counted from this AD effective date.</p> <p>2. CDCCL</p> <ul style="list-style-type: none"> - It is the responsibility of the operator to ensure that their internal documentation is amended to reflect the data contained within <i>AIRBUS</i> ALS Part 5, Fuel Airworthiness Limitations Section 2 and to provide appropriate text to highlight the existence of each CDCCL. The operators internal procedures and documentation ensuring management of control of CDCCL shall be fully implemented before 01 July 2007.

	<ul style="list-style-type: none"> - No retroactive action on aircraft in service is required further to the above mentioned amendment of the documentation.
Ref. Publications:	A320 Fuel Airworthiness Limitations, 95A.1931/05 Issue 1, or later approved revisions.
Remarks :	<ol style="list-style-type: none"> 1. If requested and appropriately substantiated the responsible EASA manager for the related product has the authority to accept Alternative Methods of Compliance (AMOCs) for this AD. 2. This AD was posted as PAD 06-006R1 for consultation on 07 June 2006 with a comment period until 22 June 2006. PAD 06-006R1 has been issued to endorse comments received for PAD 06-006 and due to the change of the EASA policy statement on fuel tank safety on March 2006. The Comment Response Document can be found at http://ad.easa.eu.int/ . 3. Enquiries regarding this Airworthiness Directive should be referred to Mr. M. Capaccio, Airworthiness Directive Focal Point - Certification Directorate, EASA. E-mail: ADs@easa.europa.eu . 4. For any question concerning the technical content of the requirements in this AD, please contact AIRBUS EAS - Fax +33 5 61 93 44 51