


<b>EASA</b>	<b>AIRWORTHINESS DIRECTIVE</b>	
	<p><b>AD No.: 2006 – 0171 R1</b></p> <p><b>Date: 14 September 2006</b></p>	
<p>No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise agreed with the Authority of the State of Registry.</p>		
<p><b>Type Approval Holder's Name:</b></p> <p>Schweizer Aircraft Corporation</p>	<p><b>Type/Model designation(s):</b></p> <p>Schweizer (Hughes) 269 Series</p>	
<p>TCDS Number: United States of America (FAA) 4H12</p>		
<p>Foreign AD: None</p>		
<p>Supersedure: EASA AD 2006-0171 dated 19 June 2006, which superseded United Kingdom CAA Additional AD 002–02–2000 Revision 1</p>		
<b>ATA 28</b>	<b>Fuel – Fuel Vent System – Modification</b>	
<p><b>Manufacturer(s):</b></p>	<p>Schweizer Aircraft Corporation; Hughes Helicopters, Inc.</p>	
<p><b>Applicability:</b></p>	<p>All 269A, 269A-1, 269B, 269C and 269C-1 helicopters.</p>	
<p><b>Reason:</b></p>	<p>To prevent release of fuel from the fuel tank vent in the event of a roll-over on landing the helicopter and the subsequent risk of fire, this Airworthiness Directive requires the modification of the Fuel Vent System.</p> <p>Note : United Kingdom AAIB Safety Recommendation 95-12 refers.</p> <p>This AD has been revised to correct the applicability (the requirement does not apply to 269D helicopters), the compliance statement and the referenced TC holder's service letter.</p>	
<p><b>Effective Date:</b></p>	<p>24 September 2006</p>	
<p><b>Compliance:</b></p>	<p>Within the next 1,000 hours time-in-service after 03 July 2006 [the effective date of the original issue of this AD] but not later than 03 July 2008, whichever occurs first, modify the helicopter's Fuel Vent System by incorporating the applicable Schweizer Aircraft Corporation Modification Kit as listed in the referenced Schweizer Service Letter L-169.</p>	
<p><b>Ref. Publications:</b></p>	<p>Schweizer Service Letter L-169 dated 10 August 2000.</p>	

Remarks:	<ol style="list-style-type: none"><li>1. If requested and appropriately substantiated the responsible EASA manager for the related product has the authority to accept Alternative Methods of Compliance (AMOCs) for this AD.</li><li>2. This AD was posted as PAD 06-101 for consultation on 19 April 2006 with a comment period until 15 May 2006. No comments were received during the consultation period.</li><li>3. Enquiries regarding this AD should be referred to the AD Focal Point - Certification Directorate, EASA. e-mail <a href="mailto:ads@easa.eu.int">ads@easa.eu.int</a>.</li><li>4. For questions concerning the technical contents of this AD requirements, contact: Schweizer Aircraft Corporation, P.O. Box 147, Elmira, New York, 14902, United States of America; telephone +1-607-739-3821; facsimile +1-607-796-2488; e-mail <a href="mailto:schweizer@sacusa.com">schweizer@sacusa.com</a>.</li></ol>
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