

PŘÍKAZ K ZACHOVÁNÍ LETOVÉ ZPŮSOBILOSTI

CAA-AD-070/2002

Datum vydání: 04. července 2002

LETOUN - NÁPOROVÁ TURBÍNA - KONTROLA/OPRAVA

Týká se: letadel AIRBUS A310, všech certifikovaných verzí a všech výrobních čísel, která nebyla zkontrolována v souladu s "AIRBUS INDUSTRIE Service Bulletin (SB) A310-57-2084" v originálním vydání nebo v jeho pozdějších revizích.

Datum účinnosti: 08. srpna 2002

Provést v termínech: Jak je popsáno v DGAC AD 2002-263(B), od data účinnosti tohoto PZZ.

Postup provedení prací: Dle DGAC AD 2002-263(B) (příloha tohoto PZZ).

Poznámky: Provedení tohoto PZZ musí být zapsáno do letadlové knihy. Případné dotazy týkající se tohoto PZZ adresujte na ÚCL sekce technická - Ing. Toman. Pokud to vyžaduje povaha tohoto PZZ, musí být zapracován do příslušné části dokumentace pro obsluhu, údržbu a opravy letadla. Tento PZZ byl vypracován na základě DGAC AD 2002-263(B).

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Úřad pro civilní letectví

DGAC AD No.: 2002-263(B)
AIRBUS

A310 aircraft

RAT (Ram Air Turbine) ejection jack gimbals assembly (ATA 29, 57)

APPLICABILITY:

AIRBUS A310 aircraft, all certified models and all serial numbers which were not inspected in accordance with AIRBUS INDUSTRIE Service Bulletin (SB) A310-57-2084, original issue or any later approved revision.

REASONS:

An A310 aircraft operator reported during a scheduled check that the two half fittings holding the RAT ejection jack were found cracked at the two forks located at the RH and LH ends of the fittings, along the stiffener. Investigation has shown that these cracks were due to stress corrosion. Conductivity test revealed that the heat treatment of the half fittings aluminum alloy was T6 in place of T7351. This incorrect heat treatment decreases drastically the material behaviour against stress corrosion and was identified as the cause of the damage.

This situation, if not corrected, could lead to a decreased structural integrity of the gimbal half fittings and to the loss of the RAT during its extension.

COMPLIANCE:

At latest within 600 flight hours following the effective date of this Airworthiness Directive, unless already accomplished, inspect the two RAT gimbal half fittings in accordance with the instructions of SB A310-57-2084. Depending on the results of this inspection, perform the potential corrective actions as per above-mentioned SB.

Note: Once the inspection and corrective action have been performed in accordance with SB A310-57-2084, no additional or repetitive inspections are required.

In case of replacement of the half fittings is necessary, the new half fittings should be P/N A572102380000 and should either have been ordered after November 2001 or should pass successfully the conductivity test as prescribed per SB A310-57-2084.

REF.: AIRBUS INDUSTRIE SB A310-57-2084 (Any later approved revision of this SB is acceptable)

EFFECTIVE DATE:

MAY 25, 2002