EASA

EMERGENCY AIRWORTHINESS DIRECTIVE



AD No.: 2011-0061-E

Date: 01 April 2011

Note: This Emergency Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation

This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].

Type Approval Holder's Name: Costruzioni Aeronautiche TECNAM S.r.l.		Type/Model designation(s): P2002-JR aeroplanes
Foreign AD :	Not Applicable	
Supersedure :	None	
ATA 32	Landing Gear– Eme Extension – Inspec	ergency Accumulator for Landing Gear tion / Replacement
Manufacturer(s):	Costruzioni Aeronautiche TECNAM S.r.l.	
Applicability:	P2002-JR aeroplanes, serial numbers: 015, 048, 052, 065, 066, 076, 09 092, 093, 110, 111, 112, 116 and 123.	
Reason:	flight inspection of a Cos aeroplane. This damage accumulator for the exte from the emergency acc	the fuselage aft tail cone was found during a prestruzioni Aeronautiche TECNAM P2006T was caused by the lower lid of the emergency ension of the landing gear. The lid had detached cumulator, violently hitting the lower skin of the damaging the accumulator cylinder.
		ected and corrected, could impair the aeroplane eopardize the landing gear emergency extension in normal mode.
	P2006T aeroplanes, and accumulator part number P2002-JR aeroplanes, the accumulator cylinder for	d reasons EASA issued AD 2011-0059-E on d considering that the same emergency or (P/N) 22-9-610-000 is also installed on his AD requires an inspection of the emergency absence of crack, deformation or oil leakage, a he applicable corrective actions.
		be an interim measure and, after approval of a signed by the Type Certificate holder, further AD

EASA Form 111 Page 1/2

Effective Date:	05 April 2011	
Required action(s) and Compliance Time(s):	Required as indicated, unless previously accomplished: (1) Before next flight after the effective date of this AD, inspect the emergency accumulator P/N 22-9-610-000 for absence of crack, deformation or oil leakage, in accordance with the instructions of Costruzioni Aeronautiche TECNAM Service Bulletin N. SB 049–CS. (2) If a crack, deformation or oil leakage is found during the inspection required by paragraph (1) of this AD, before next flight replace the emergency accumulator with a serviceable part, in accordance with the relevant instructions of the Aircraft Maintenance Manual.	
Ref. Publications:	Costruzioni Aeronautiche TECNAM Service Bulletin N. SB 049-CS dated 29 March 2011. The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.	
Remarks:	 If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD. The safety assessment has requested not to implement the full consultation process and an immediate publication and notification. Enquiries regarding this AD should be referred to the Airworthiness Directives, Safety Management & Research Section, Certification Directorate, EASA. E-mail: ADs@easa.europa.eu. For any question concerning the technical content of the requirements in this AD, please contact: Costruzioni Aeronautiche TECNAM Airworthiness Office; Phone: +39 0823 620134; fax: +39 0823 622899; E-mail: m.oliva@tecnam.com; p.violetti@tecnam.com. 	

EASA Form 111 Page 2/2