

EASA	EMERGENCY AIRWORTHINESS DIRECTIVE	
	EAD No.: 2010-0037-E	
	<p>Date: 08 March 2010</p> <p>Note: This Emergency Airworthiness Directive (EAD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation</p>	
<p>This EAD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].</p>		
<p>Type Approval Holder's Name : Aircraft Industries, a.s.</p>	<p>Type/Model designation(s) : L 23 SUPER-BLANÍK sailplanes</p>	
<p>TCDS Number : EASA.A.044</p>		
<p>Foreign AD : Not applicable</p>		
<p>Supersedure : None</p>		
ATA 55	Stabilizers - Elevator Inner Hinges – Inspection / Replacement	
<p>Manufacturer(s):</p>	<p>Aircraft Industries, a.s, LET k.p., LET a.s. and LETECKÉ ZÁVODY a.s.</p>	
<p>Applicability:</p>	<p>L 23 SUPER-BLANÍK sailplanes, all serial numbers.</p>	
<p>Reason:</p>	<p>Cracks on the stabilizer elevator inner hinges of seven L 23 SUPER-BLANÍK sailplanes have been detected during an inspection.</p> <p>This condition, if not corrected, could result in no longer retaining the elevator in place and in jamming of the Pilot's elevator control system, and subsequent loss of elevator control.</p> <p>For the reasons stated above, this Emergency AD requires the inspection of the elevator inner hinges, and the accomplishment of the relevant corrective actions as necessary.</p>	
<p>Effective Date:</p>	<p>10 March 2010</p>	
<p>Required action(s) and Compliance Time(s):</p>	<p>Required as indicated:</p> <p>(1) Before next flight after the effective date of this Emergency AD, inspect the elevator inner hinges on the stabilizer in accordance with paragraphs A1, A2 and A4 of the Aircraft Industries, a.s. Mandatory Bulletin (MB) L23/052a.</p>	

	<p>(2) Following the inspection required by paragraph (1) of this Emergency AD, if any elevator inner hinges on the elevator is cracked or damaged, before next flight, replace it in accordance with paragraphs A3 and A4 of the Aircraft Industries, a.s. MB L23/052a.</p> <p>(3) Repeat the inspection defined in paragraph (1) of this Emergency AD at intervals not exceeding 1000 Flight Hours from the last inspection and apply the associated corrective actions as defined in paragraph (2) of this Emergency AD, if necessary.</p>
Ref. Publications:	<p>Aircraft Industries, a.s. Mandatory Bulletin No. L23/052a, original issue, dated 02 March 2010.</p> <p>The use of later approved revisions of this document is acceptable for compliance with the requirements of this EAD.</p>
Remarks :	<ol style="list-style-type: none"> 1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this EAD. 2. The safety assessment has requested not to implement the full consultation process and an immediate publication and notification. 3. Enquiries regarding this EAD should be referred to the Airworthiness Directives, Safety Management & Research Section, Certification Directorate, EASA. E-mail: ADs@easa.europa.eu. 4. For any question concerning the technical content of the requirements in this EAD, please contact: Aircraft Industries, a.s. - Na záhonech 1177, 686 04 Kunovice, Czech Republic Telephone: +420 572 817 660, Fax: +420 572 816 112 Email: ots@let.cz