

ÚŘAD PRO CIVILNÍ LETECTVÍ

SEKCE TECHNICKÁ

PŘÍKAZ K ZACHOVÁNÍ LETOVÉ ZPŮSOBILOSTI

Číslo: 2009-0036

Datum účinnosti: 06. března 2009

CFM International SA (CFM)

CFM56-3

Tento PZZ je vydáván pro výrobek transferovaný pod působnost EASA.

Na základě rozhodnutí EASA je následující Příkaz k zachování letové způsobilosti závazný pro všechny výrobky provozované v EU na které se daný PZZ vztahuje.

Provedení PZZ, který se vztahuje podle typu a výrobního čísla na výrobek je pro provozovatele/vlastníka letadla zapsaného do leteckého rejstříku závazné. Neprovedením PZZ ve stanoveném termínu dojde ke ztrátě letové způsobilosti výrobku.

Poznámky

- Provedení tohoto PZZ musí být zapsáno do provozní dokumentace letadla.
- Případné dotazy týkající se tohoto PZZ adresujte na ÚCL sekce technická.
- Pokud to vyžaduje povaha tohoto PZZ, musí být zapracován do příslušné části dokumentace pro obsluhu, údržbu a opravy letadla.

EASA AD No: 2009-0036

EASA

AIRWORTHINESS DIRECTIVE



AD No.: 2009-0036

Date: 20 February 2009

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008 Article 14(4) exemption]

cle 14(4) exemption].	
older's Name :	Type/Model designation(s):
SA (CFM)	CFM56-3 turbofan engines
EASA.E.066	
Not applicable	
None	
Г	
Engine – Fan B	Blades with 25 Degree Mid-span Shrouds – Inspections
Snecma, General Elec	tric
CFM International (CFM) CFM56-3 turbofan engines, all serial numbers, when equipped with fan blades - with 25 degree mid-span shrouds - Part Number (P/N) 9527M99P08, 9527M99P09, 9527M99P10, 9527M99P11, an 1285M39P01, or fan blade pairs P/N 335-088-901-0, 335-088-902-0, 335-088-903-0, and 335-088-904-0.	
These engines are kno 737series aeroplanes.	own to be installed on, but not limited to, Boeing
Remark: The installation not allowed on the CFM	on of fan blades with 25 degree mid-span shrouds is M56-3C engine model.
Severe wear was observed contact location with the	blade failure has been reported on a CFM56-3 engine. ved on the underside of the fan blade platforms at the fan blade dampers. This resulted in poor damping of the buted to high dynamic stress of the blades.
blades with 25 degrees CFM56-3/3B/3C S/B 72-	above, this AD requires repetitive inspections of fan s mid-span shrouds as defined by CFM service bulletin -1067 in order to reduce the risk of a multiple fan blade d result in an uncontained engine rotor failure.
	Snecma, General Electors CFM International (CFI when equipped with fair Number (P/N) 9527M9 1285M39P01, or fan blows 1285M39P01, and 335-08 These engines are known 737series aeroplanes. Remark: The installation not allowed on the CFM An event of multiple fan Severe wear was observed to cation with the fan blade set and contributed for the reasons stated a blades with 25 degrees CFM56-3/3B/3C S/B 725

EASA Form 110 Page 1/2

Effective Date:	06 March 2009	
Required Action(s) and Compliance Time(s):	(1) Within 3 months after the effective date of this AD, perform an inspection of fan blades P/N 9527M99P08, 9527M99P09, 9527M99P10, 9527M99P11, and 1285M39P01, or fan blade pairs P/N 335-088-901-0, 335-088-902-0, 335-088-903-0, and 335-088-904-0 in accordance with the accomplishment instructions of paragraphs 3.A (On-wing) or 3.B (In-shop) of CFM Service Bulletin (SB) CFM56-3/3B/3C 72-1067 original issue.	
	If any damage is found, before further flight, remove and replace the blades as instructed in the accomplishment instructions of CFM SB CFM56-3/3B/3C S/B 72-1067 original issue.	
	(2) Accomplishment of inspections done in accordance with CFM Service Bulletin (SB) CFM56-3/3B/3C S/B 72-1067 original issue, within the last 3 000 engine Flight Cycles (FC) before the effective date of this AD satisfies the requirements of paragraph (1) of this AD.	
	(3) Thereafter, at intervals not to exceed 3 000 engine FC from the last inspection, repeat the inspections required in paragraph (1) of this AD.	
	(4) After the affective date of this AD, do not install on any CFM56-3 engine fan blades P/N 9527M99P08, 9527M99P09, 9527M99P10, 9527M99P11, and 1285M39P01, or fan blade pairs P/N 335-088-901-0, 335-088-902-0, 335-088-903-0, and 335-088-904-0 unless they have been previously inspected in accordance with paragraph 3 of the accomplishment instructions of CFM SB CFM56-3/3B/3C 72-1067 original issue.	
	CFM service bulletin CFM56-3/3B/3C S/B 72-1067, original issue.	
Ref. Publications:	The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.	
Remarks :	If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.	
	 This AD was posted on 27 January 2009 as PAD 09-025 for consultation until 17 February 2009.No comments were received during the consultation period. 	
	 Enquiries regarding this AD should be referred to the Airworthiness Directives, Safety Management & Research Section, Certification Directorate, EASA. E-mail ADS@easa.europa.eu. 	
	4. For any questions concerning the technical content of the requirements in this AD, please contact: CFM SA Customer Support Center E-mail: snecma.csc@snecma.fr International Phone: +33 1 64 14 88 66 Fax: +33 1 64 79 85 55 or CFM Inc. Customer Support Center E-mail: geae.csc@ae.ge.com International Phone: +1 513-552-3272 USA Phone: 877-GEAE-CSC	

EASA Form 110 Page 2/2