EASA	AIRWORTHINESS DIRECTIVE		
1	AD No.: 2008-0218		
	Date: 10 December 2008		
E	Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.		
This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].			
Type Approval Holder's Name :		Type/Model designation(s) :	
ATR-GIE Avions de Transport Régional		ATR 42 and ATR 72 aircraft	
TCDS Number : EASA.A.084			
Foreign AD : Not applicable			
Supersedure : None			
ATA 92	Electrical / Electronic - Wire Bundles in the Rear Baggage Zone - Protection and Clamping		
Manufacturer(s):	ATR-GIE Avions de Transport Régional (formerly AEROSPATIALE – AERITALIA, AEROSPATIALE – ALENIA, AEROSPATIALE ATR – ALENIA, EADS ATR – ALENIA)		
Applicability:	ATR 42-500 aircraft, with Manufacturer Serial Numbers (MSN) 667, 669 and 671 and		
	ATR 72-212A aircraft from MSN 756 to 784 inclusive, except MSN 770, 773 and 783. NOTE: MSN 770, 773 and 783 have already been inspected prior to delivery by ATR.		
Reason:	Identification of an unsafe condition related to the loss of the fail-safe design criteria for the rudder trim, pitch trim and stick pusher control functions after a reported event led in April 2008 to the release of AD 2008-0062 to mandate the ATR modification No.05780.		
	It has appeared that some aircraft manufactured and delivered before the 1 st of April 2008 may have received on the production-line a partial or incorrect implementation of the required ATR mod. No.05780.		
	the identified aircraft to	above, the present AD requires an inspection of verify and, finalize when necessary, the complete ation of the ATR mod. No.05780	

Effective Date:	24 December 2008	
Required action(s) and Compliance Time(s):	 Required as indicated unless already accomplished: (1) Within 550 Flight Hours (FH) after the effective date of this AD, verify the installation of the appropriate protection sleeves and the correct routing and clamping of the wire bundles in accordance with the accomplishment instructions of ATR Service Bulletin ATR42-92-0020 or ATR72-92-1021, as applicable to aircraft models. (2) If any discrepancy is found, before further flight, do all corrective actions as instructed in the ATR Service Bulletin ATR42-92-0020 or ATR72-92-1021, as applicable to aircraft models. 	
Ref. Publications:	Service Bulletin ATR42-92-0020 original issue Service Bulletin ATR72-92-1021 original issue The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.	
Remarks :	 If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD. This AD was posted on 05 November 2008 as PAD 08-125 for consultation until 26 November 2008. No comments were received during the consultation period. Enquiries regarding this AD should be referred to the Airworthiness Directives, Safety Management & Research Section, Certification Directorate, EASA. E-mail <u>ADs@easa.europa.eu</u>. For any questions concerning the technical content of the requirements in this AD, please contact: ATR - GIE Avions de Transport Régional, Continued Airworthiness Service, Tel.: +33 (0)5 62 21 62 21 - Fax: +33 (0) 5 62 21 67 18; E-mail: continued.airworthiness@atr.fr 	