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## PŘÍKAZ K ZACHOVÁNÍ LETOVÉ ZPŮSOBILOSTI

**Číslo: 2008-0103** 

Datum účinnosti: 12. června 2008

Aircraft Industries a.s.

LET L410UVP

Tento PZZ je vydáván pro výrobek transferovaný pod působnost EASA

Na základě rozhodnutí EASA je následující Příkaz k zachování letové způsobilosti závazný pro všechny výrobky provozované v EU, na které se daný PZZ vztahuje.

Provedení PZZ, který se vztahuje podle typu a výrobního čísla na výrobek je pro provozovatele/vlastníka letadla zapsaného do leteckého rejstříku závazné. Neprovedením PZZ ve stanoveném termínu dojde ke ztrátě letové způsobilosti výrobku.

## Poznámky.

<sup>-</sup> Provedení tohoto PZZ musí být zapsáno do provozní dokumentace letadla.

<sup>-</sup> Případné dotazy týkající se tohoto PZZ adresujte na ÚCL sekce technická.

<sup>-</sup> Pokud to vyžaduje povaha tohoto PZZ, musí být zapracován do příslušné části dokumentace pro obsluhu, údržbu a opravy letadla.

## EASA

## **AIRWORTHINESS DIRECTIVE**

AD No.: 2008-0103

Date: 29 May 2008

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].

Type Approval Holder's Name :		Type/Model designation(s) :	
Aircraft Industries a.s.		LET L410UVP	
TCDS Number: EA	ASA A.026		
Foreign AD : No	ot applicable		
Supersedure: No	one		
ATA: Multiple	Introduction of Cab	in and Aircraft Systems Safety Modifications	
Manufacturer(s):	Aircraft Industries a.s.	(formerly known as Letecké Závody a.s. alias LET)	
Applicability:	All L410UVP Turbolet a	aircraft	
Reason:	them from being certification these aircraft were trans	ET L410 series had certification bases that precluded cated in EU member states. On accession to the EU, asferred on the basis that essential safety e introduced to enable continuing operation in EU	
		mprovements is to bring the aircraft to a safety other similar types certificated in EU member states.	
	In effect, analysis of the LET-410 safety records has revealed a higher accident rate than other types of similar vintage undertaking similar operations. To mitigate this, it is necessary to improve the cabin safety of the LET-410 aeroplanes to optimize the passenger evacuation in case of emergency and to incorporate a series of aircraft systems modifications which, if not implemented, could lead to unsafe conditions that would result in safety of flight issues. Consequently, this Airworthiness Directive (AD) mandates embodiment of several modifications which consist of two separate sets:		
	Part A "Additional ememodifications"	ergency exits installation under wings and related	
	<u>and</u>		

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	Part B: Remaining minimum essential safety modifications		
	For details refer to the compliance block		
Effective Date:	12 June 2008		
Required action(s) and Compliance Time(s):	Before 01 July 2010, Modify the airplane in accordance with Aircraft Industries a.s. Information Bulletin IB No. L410UVP/149b to embody either Part A and Part B or Part B solely of the following sets of compulsory modifications as listed below:		
	Part A: Additional emergency exits installation under wings on left-hand fuselage side and related modifications:		
	1.Modification of fuselage structure 2.Installation of emergency exits 3.Modification of fuselage insulation and upholstery 4.Anti-skid carpeting on landing gear nacelles 5.Emergency path lighting 6.Emergency exit signs and placards 7.Emergency lighting wiring 8.Other related modifications		
	Part B: Remaining minimum essential safety modifications:		
	<ul> <li>1.Modification of locking mechanism of the front emergency exit</li> <li>2.Installation of indication of closing of the front emergency exit and covers of the front baggage compartment</li> <li>3.Installation of handle at front emergency exit</li> <li>4.Installation of labels on front emergency exit</li> <li>5.Installation of temperature indicator of onboard storage batteries</li> <li>6.Installation of non-return valves in wing fuel tank filler necks and elimination of internal fuel filler necks on wing</li> <li>7.Installation of fuel flow meters</li> <li>8.Installation of warning device of exceeding maximum operating speed V<sub>MO</sub></li> <li>9.Re-routing of rudder and aileron trim tab cables</li> <li>10.Installation of protective fire-resistant coating on hoses in engine nacelle area</li> <li>11.Modification of passenger seats to prevent shifting of baggage in under-seat area</li> <li>12.Installation of water collector into pressure air piping of de-icing system of tail units</li> <li>13.Modification of fuselage harness</li> <li>14.Replacement of rudder blocking device</li> </ul>		
	The modifications are to be implemented by Aircraft Industries a.s. although alternative methods of compliance may be accepted (see Remark 1 below).		
	FOR AIRCRAFT MODIFIED IN ACCORDANCE WITH PARTS A AND B, THERE ARE NO ADDITIONAL LIMITATIONS beyond those listed in Type Certificate Data Sheet EASA.A.026.		
	AIRCRAFT MODIFIED IN ACCORDANCE WITH PART B ONLY WILL BE LIMITED TO A MAXIMUM OF NINE PASSENGERS after 30 June 2010. (This restriction does not apply to aircraft used in parachuting operations.)		
	NOTE: These limitations are set out in Type Certificate Data Sheet EASA.A.026, taking effect on the compliance date of this AD.		
Ref. Publications:	Aircraft Industries a.s. Information Bulletin IB No. L410UVP/149b		

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Remarks	
Nemano	

- 1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
- 2. This AD was posted on 15 February 2008 as PAD 08-021 for consultation until 14 March 2008. The Comment Response Document can be found at <a href="http://ad.easa.europa.eu">http://ad.easa.europa.eu</a>.
- 3. Enquiries regarding this AD should be referred to the Airworthiness Directives, Safety Management & Research Section, Certification Directorate, EASA. E-mail ADs@easa.europa.eu
- 4. For any questions concerning the technical content of the requirements in this AD, please contact:

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