

# PŘÍKAZ K ZACHOVÁNÍ LETOVÉ ZPŮSOBILOSTI

**CAA-AD-097/1999**

Datum vydání: 20. září 1999

## LETADLO - ZDROJ ELEKTRICKÉ ENERGIE - KONTROLA/VÝMĚNA

**Týká se:** všech letadel Boeing 737-100, -200, -300, -400 a -500, certifikovaných v kterékoliv kategorii.

**Datum účinnosti:** 04. listopad 1999

**Provést v termínech:** Jak je popsáno v FAA AD 99-18-17 (příloha tohoto PZZ).

**Postup provedení prací:** Dle FAA AD 99-18-17.

Poznámky: Provedení tohoto PZZ musí být zapsáno do letadlové knihy. Případné dotazy týkající se tohoto PZZ adresujte na ÚCL technický inspektorát - Ing. Toman. Pokud to vyžaduje povaha tohoto PZZ musí být zapracován do příslušné části dokumentace pro obsluhu, údržbu a opravy letadla. Tento PZZ byl vypracován na základě FAA AD 99-18-17.

**Ing. Pavel MATOUŠEK**  
**Ředitel technického inspektorátu**  
**Úřad pro civilní letectví**

99-18-17 BOEING: Amendment 39-11283. Docket 99-NM-187-AD. Issued August 24, 1999.

Applicability: All Model 737-100, -200, -300, -400, and -500 series airplanes; certificated in any category.

NOTE 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (f) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent failure of all electrically powered airplane systems, which could result in the inability to continue safe flight and landing, accomplish the following:

(a) For Model 737-100 and -200 series airplanes equipped with battery charger Boeing part number (P/N) 10-60701-1: Within 90 days after the effective date of this AD, accomplish the requirements of paragraphs (a)(1) and (a)(2) of this AD.

(1) Replace the airplane battery charger with a new or serviceable airplane battery charger, Boeing P/N 10-60701-3, in accordance with Chapter 20-10-111 of the Boeing 737 Airplane Maintenance Manual (AMM); and

(2) Replace the airplane battery with a new or reconditioned airplane battery in accordance with Chapter 24-31-11 of the Boeing 737 AMM. Thereafter, replace the airplane battery with a new or reconditioned airplane battery at intervals not to exceed 750 flight hours.

(b) For Model 737-300, -400, and -500 series airplanes: Within 90 days after the effective date of this AD, replace the airplane battery with a new or reconditioned airplane battery in accordance with Chapter 24-31-11 of the Boeing 737 AMM. Thereafter, replace the airplane battery with a new or reconditioned airplane battery at intervals not to exceed 750 flight hours.

(c) For all airplanes: Within 90 days after the effective date of this AD, perform a test to determine the condition of diode CR910 of the Generator Control Units (GCU) in accordance with Boeing Telex Message M-7200-99-01528, dated March 5, 1999.

NOTE 2: Any tests performed prior to the effective date of this AD, in accordance with Boeing Telex Message M-7200-99-01528, dated February 19, 1999, or dated March 4, 1999, are not considered acceptable for compliance with the applicable action specified by this AD.

(1) If all diodes pass the test, repeat the diode test thereafter, at intervals not to exceed 600 flight hours.

(2) If any diode fails the test: Prior to further flight, replace the GCU with a new or serviceable GCU, and if necessary, the airplane battery with new or reconditioned airplane battery, and repeat the diode test for the replaced GCU in accordance with the telex message until successful completion of the test is achieved. Repeat the diode test thereafter, at intervals not to exceed 600 flight hours.

(d) As of the effective date of this AD, no person shall install a battery charger having P/N 10-60701-1 on any Model 737 series airplane.

(e) Within 10 days after accomplishing the initial diode test required by paragraph (c) of this AD, submit a report of the test results (negative findings) to the Manager, Seattle Aircraft Certification Office (ACO), FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; fax (425) 227-1181. Information collection requirements contained in this regulation have been approved by the Office of Management and Budget (OMB) under the provisions of the Paperwork Reduction Act of 1980 (44 U.S.C. 3501 et seq.) and have been assigned OMB Control Number 2120-0056.

#### Alternative Methods of Compliance

(f) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Seattle ACO. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Seattle ACO.

NOTE 3: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Seattle ACO.

#### Special Flight Permits

(g) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

#### Incorporation by Reference

(h) Except as provided by paragraphs (a) and (b) of this AD, the test shall be done in accordance with Boeing Telex Message M-7200-99-01528, dated

March 5, 1999. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Boeing Commercial Airplane Group, P.O. Box 3707, Seattle, Washington. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

(i) This amendment becomes effective on September 16, 1999.

FOR FURTHER INFORMATION CONTACT:

Stephen S. Oshiro, Aerospace Engineer, Systems and Equipment Branch, ANM-130S, FAA, Transport Airplane Directorate, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2793; fax (425) 227-1181.