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PŘÍKAZ K ZACHOVÁNÍ LETOVÉ ZPŮSOBILOSTI

Číslo: CAA-AD-088/2003

Datum vydání: 19. listopadu 2003

CESSNA AIRCRAFT COMPANY
560

LETOUN - SYSTÉM PODÉLNÉHO VYVÁŽENÍ - KONTROLA

Týká se: letadel Cessna 560, majících výrobní číslo 0260 až 0396 včetně, certifikovaných v kterékoliv kategorii; mimo letadel, na kterých byl proveden Cessna Service Bulletin 560-34-93, vydaný 16. března 2001.

Datum účinnosti: 29. prosince 2003.

Provést v termínech:

Jak je popsáno v FAA AD 2003-23-02 od data účinnosti tohoto PZZ.

Postup provedení prací:

Dle FAA AD 2003-23-02 (příloha tohoto PZZ).

Poznámky:

- Provedení tohoto PZZ musí být zapsáno do letadlové knihy.
- Případné dotazy týkající se tohoto PZZ adresujte na ÚCL sekce technická – Ing. Šorm.
- Pokud to vyžaduje povaha tohoto PZZ, musí být zapracován do příslušné části dokumentace pro obsluhu, údržbu a opravy letadla.
- Tento PZZ byl vypracován na základě FAA AD 2003-23-02.

Ing. Pavel MATOUŠEK
ředitel

2003-23-02 Cessna Aircraft Company: Amendment 39-13365. Docket 2003- NM-225-AD.

Applicability: Model 560 airplanes, having serial numbers 0260 through 0396 inclusive, certificated in any category; except those on which Cessna Service Bulletin 560-34-93, dated March 16, 2001, has been accomplished.

Compliance: Required as indicated, unless accomplished previously.

To prevent a single-point failure in the trim system from causing a runaway trim condition that the pilot may be unable to stop by using the autopilot disconnect switch, which could result in loss of control of the airplane, accomplish the following:

Disengaging and Tie-Strapping Circuit Breakers

(a) Within 5 days or 10 hours time-in-service after the effective date of this AD, whichever is first: Disengage the PITCH TRIM circuit breaker on the left circuit breaker panel and the SERVO 1 circuit breaker on the right circuit breaker panel, and install tie straps on those circuit breakers, per paragraphs 1.A. and 1.B. of the Accomplishment Instructions of Cessna Alert Service Letter ASL560-27-10, dated October 10, 2003.

Optional Inspection and Corrective Actions

(b) Accomplishment of the inspection of the trim pc board assembly to determine the part number of the assembly and all applicable follow-on actions; per paragraphs 2.A., 2.B., and 2.C. of the Accomplishment Instructions of Cessna Alert Service Letter ASL560-27-10, dated October 10, 2003; terminates the requirements of paragraph (a) of this AD. Once the inspection and applicable follow-on actions have been accomplished, the tie straps on the pitch trim and autopilot servo circuit breakers may be removed and those circuit breakers may be re-engaged.

Parts Installation

(c) As of the effective date of this AD, no person may install a trim pc board assembly having part number 6518351-3 or -5 on any airplane.

Alternative Methods of Compliance

(d) In accordance with 14 CFR 39.19, the Manager, Wichita Aircraft Certification Office (ACO), FAA, is authorized to approve alternative methods of compliance for this AD.

Incorporation by Reference

(e) The actions shall be done in accordance with Cessna Alert Service Letter ASL560-27-10, dated October 10, 2003. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Cessna Aircraft Co., P.O. Box 7706, Wichita, Kansas 67277. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; at the FAA, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Mid-Continent Airport, Wichita, Kansas; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Effective Date

(f) This amendment becomes effective on November 28, 2003.

▼ Footer Information

Issued in Renton, Washington, on November 4, 2003.

Ali Bahrami,

Acting Manager, Transport Airplane Directorate,
Aircraft Certification Service.

[FR Doc. 03-28166 Filed 11-12-03; 8:45 am]

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