

PŘÍKAZ K ZACHOVÁNÍ LETOVÉ ZPŮSOBILOSTI

CAA-AD-066/1999R1

Nahrazuje CAA-AD-066/1999

Datum vydání: 25. června 1999

LETADLO - UCHYCENÍ SPOILERU č. 3 (ATA 57) - KONTROLA

Týká se: letadel Airbus A310 všech verzí a všech výrobních čísel, jak je uvedeno v části "Applicability" DGAC AD 98-483-271(B) R1.

Datum účinnosti: 12. srpna 1999

Provést v termínech: Jak je popsáno v DGAC AD 98-483-271(B) R1.

Postup provedených prací: Dle DGAC AD 98-483-271(B) R1.

Poznámky: Provedení tohoto PZZ musí být zapsáno do letadlové knihy. Případné dotazy týkající se tohoto PZZ adresujte na ÚCL technický inspektorát - Ing. Toman. Pokud to vyžaduje povaha tohoto PZZ musí být zapracován do příslušné části dokumentace pro obsluhu, údržbu a opravy letadla. Tento PZZ byl vypracován na základě DGAC AD 98-483-271(B) R1, který nahrazuje DGAC AD 1998-483-271(B).

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Úřad pro civilní letectví

DGAC AD 1998-483-271(B) R1

AIRBUS INDUSTRIE

A310 Aircraft

Spoiler No. 3 - Actuator attachment fitting (ATA 57)

1. APPLICABILITY:

AIRBUS INDUSTRIE A310 aircraft, all certified models and all serial numbers on which AIRBUS INDUSTRIE modifications No. 04117 or 04799 have been embodied in production but on which AIRBUS INDUSTRIE modification No. 11929 (AIRBUS INDUSTRIE Service Bulletin A310-57-2079) has not been embodied.

2. REASONS:

Routine inspections have revealed the presence of cracks in spoiler No. 3 actuator attachment fittings.

The propagation of such cracks could result in the deformation of the rear spar web and the associated titanium reinforcing doubler; this could :

- generate fuel leaks,
- lead to the loss of various hydraulic and electrical systems.

This revision 1 is an up dating of the applicability definition.

3. ACTIONS AND COMPLIANCE TIME:

a) Prior to the accumulation of 10,800 flights, or within 800 flights following the effective date of the original issue of this Airworthiness Directive, whichever occurs later, perform a High Frequency Eddy Current (HFEC) inspection and take corrective actions if required, in accordance with the instructions of AIRBUS INDUSTRIE Service Bulletin A310-57-2078.

Aircraft which have accumulated more than 14,200 flights on the effective date of this Airworthiness Directive must be inspected within the following 400 flights.

Aircraft which have accumulated more than 15,400 flights on the effective date of the original issue of this Airworthiness Directive must be inspected within the following 200 flights.

b) Repeat the same inspection process every 1,200 flights or modify spoiler No. 3 attachment fittings in accordance with the instructions of AIRBUS INDUSTRIE Service Bulletin A310-57-2079.

No further actions in accordance with this Airworthiness Directive are required after accomplishment of AIRBUS INDUSTRIE Service Bulletin A310-57-2079.

Inform the manufacturer of all inspection findings whichever they are.

This Revision 1 replaces the Airworthiness Directive 1998-483-271 (B) dated December 02, 1998.

REF:

AIRBUS INDUSTRIE Service Bulletins

A310-57-2078 and A310-57-2079

(at their original issue or any other later approved revision).

EFFECTIVE DATES:

Original AD: DECEMBER 12, 1998

Revision 1: JUNE 12, 1999