

PŘÍKAZ K ZACHOVÁNÍ LETOVÉ ZPŮSOBILOSTI

CAA-AD-066/1999

Datum vydání: 25. června 1999

LETADLO - UCHYCENÍ SPOILERU č. 3 (ATA 57) - KONTROLA

Týká se: letadel Airbus A310 všech verzí a všech výrobních čísel, jak je uvedeno v části "Applicability" DGAC AD 98-483-271(B) (příloha tohoto PZZ).

Datum účinnosti: 12. srpna 1999

Provést v termínech: Jak je popsáno v DGAC AD 98-483-271(B).

Postup provedených prací: Dle DGAC AD 98-483-271(B).

Poznámky: Provedení tohoto PZZ musí být zapsáno do letadlové knihy. Případné dotazy týkající se tohoto PZZ adresujte na ÚCL technický inspektorát - Ing. Toman. Pokud to vyžaduje povaha tohoto PZZ musí být zpracován do příslušné části dokumentace pro obsluhu, údržbu a opravy letadla. Tento PZZ byl vypracován na základě DGAC AD 98-483-271(B).

Ing. Pavel MATOUŠEK
Ředitel technického inspektorátu
Úřad pro civilní letectví

DGAC AD 98-483-271(B)

AIRBUS INDUSTRIE

A310 Aircraft

Spoiler No3 - Actuator attachment fitting (ATA 57)

APPLICABILITY:

AIRBUS INDUSTRIE A310 aircraft, all certified models and all serial numbers on which AIRBUS INDUSTRIE modification No. 04117 has been embodied in production but on which AIRBUS INDUSTRIE modification No. 11929 (AIRBUS INDUSTRIE Service Bulletin A310-57-2079) has not been embodied.

REASONS: Routine inspections have revealed the presence of cracks in spoiler No 3 actuator attachment fittings.

The propagation of such cracks could result in the deformation of the rear spar web and the associated titanium reinforcing doubler ; this could :

- generate fuel leaks,
- lead to the loss of various hydraulic and electrical systems.

ACTIONS AND COMPLIANCE TIME:

a) Prior to the accumulation of 10,800 flights, or within 800 flights following the effective date of this Airworthiness Directive, whichever occurs later, perform a High Frequency Eddy Current (HFEC) inspection and take corrective actions if required, in accordance with the instructions of AIRBUS INDUSTRIE Service Bulletin A310-57-2078. Aircraft which have accumulated more than 14,200 flights on the effective date of this Airworthiness Directive must be inspected within the following 400 flights.

Aircraft which have accumulated more than 15,400 flights on the effective date of this Airworthiness Directive must be inspected within the following 200 flights.
b) Repeat the same inspection process every 1,200 flights or modify spoiler No. 3 attachment fittings in accordance with the instructions of AIRBUS INDUSTRIE Service Bulletin A310-57-2079.

No further actions in accordance with this Airworthiness Directive are required after accomplishment of AIRBUS INDUSTRIE Service Bulletin A310-57-2079.

Inform the manufacturer of all inspection findings whichever they are.

REF.:

AIRBUS INDUSTRIE Service Bulletins

A310-57-2078 and A310-57-2079

(at their original issue or any other later approved revision).

EFFECTIVE DATE: DECEMBER 12, 1998