



PŘÍKAZ K ZACHOVÁNÍ LETOVÉ ZPŮSOBILOSTI

ÚŘAD PRO CIVILNÍ LETECTVÍ
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Číslo: CAA-AD-042/2004 - oprava
Datum vydání 08. července 2004
CESSNA AIRPLANE Comp.
500, 501, 550, 551

Tento PZZ byl vydán na základě Rozhodnutí č. 2/2003 výkonného ředitele EASA, které ustanovuje, že PZZ vydané úřadem státu typového návrhu jsou závazné pro všechny země EU.

LETOUN – DISKY STATORU BRZDOVÉ SESTAVY – KONTROLA/VÝMĚNA

Týká se: letadel Cessna 500 a 501, výrobních čísel 0001 až 0689 včetně, a letadel 550 a 551, výrobních čísel 0002 až 0733 včetně; certifikovaných v kterékoliv kategorii, vybavených brzdovou sestavou BFGoodrich katalogového čísla (P/N) 2-1528-6 nebo 2-1530-4.

Datum účinnosti: 02. září 2004.

Provést v termínech:

Jak je popsáno v FAA AD 2004-09-05 correction od data účinnosti tohoto PZZ.

Postup provedení prací:

Dle FAA AD 2004-09-05 correction (příloha tohoto PZZ).

Poznámky:

- Provedení tohoto PZZ musí být zapsáno do letadlové knihy.
- Případné dotazy týkající se tohoto PZZ adresujte na ÚCL sekce technická – Ing. Šorm.
- Pokud to vyžaduje povaha tohoto PZZ, musí být zapracován do příslušné části dokumentace pro obsluhu, údržbu a opravy letadla.
- Tento PZZ byl vypracován na základě FAA AD 2004-09-05 correctin.

Ing. Pavel MATOUŠEK
ředitel

2004-09-05 Cessna Airplane Company: Amendment 39-13594. Docket 2000-NM-65-AD.

Applicability: Model 500 and 501 airplanes, serial numbers 0001 through 0689 inclusive, and Model 550 and 551 airplanes, serial numbers 0002 through 0733 inclusive; certificated in any category; equipped with BFGoodrich brake assembly part number (P/N) 2-1528-6 or 2-1530-4.

Compliance: Required as indicated, unless accomplished previously.

To prevent jamming of the wheel/tire assembly, which could result in a loss of directional control or braking performance upon landing, accomplish the following:

Inspection of Stator Disks for Change Letter

(a) Within 50 landings or 90 days after the effective date of this AD, whichever is first, inspect the stator disks on the brake assembly to determine if "CHG AI" or "CHG B" or a higher change letter is impression-stamped on each disk, in accordance with Goodrich Service Bulletin 2-1528-32-2 (for airplanes equipped with BFGoodrich brake assembly P/N 2-1528-6); or Goodrich Service Bulletin 2-1530-32-2 (for airplanes equipped with BFGoodrich brake assembly P/N 2-1530-4); both Revision 5; both dated February 19, 2003; as applicable. If both disks are stamped with "CHG AI" or "CHG B" or a higher change letter, no further action is required by this paragraph. A review of airplane maintenance records is acceptable in lieu of an inspection of the stator disks if the change letter of the stator disks can be positively determined from that review.

Inspection for Cracked or Broken Stator Disks

(b) For any stator disk not stamped with "CHG AI" or "CHG B" or a higher change letter: At the applicable compliance time specified in paragraph (b)(1) or (b)(2) of this AD, perform a detailed inspection for cracked or broken stator disks; in accordance with Goodrich Service Bulletin 2-1528-32-2 (for airplanes equipped with BFGoodrich brake assembly P/N 2-1528-6); or Goodrich Service Bulletin 2-1530-32-2 (for airplanes equipped with BFGoodrich brake assembly P/N 2-1530-4); both Revision 5; both dated February 19, 2003; as applicable.

(1) For airplanes that use thrust reversers: Inspect prior to the accumulation of 376 total landings on the brake assembly, or within 50 landings after the effective date of this AD, whichever is later.

(2) For airplanes that do not use thrust reversers: Inspect prior to the accumulation of 200 total landings on the brake assembly, or within 50 landings after the effective date of this AD, whichever is later.

Note 1: For the purposes of this AD, a detailed inspection is defined as: "An intensive visual examination of a specific structural area, system, installation, or assembly to detect damage, failure, or irregularity. Available lighting is normally supplemented with a direct source of good lighting at intensity deemed appropriate by the inspector. Inspection aids such as mirror, magnifying lenses, etc., may be used. Surface cleaning and elaborate access procedures may be required."

Follow-On Actions (No Cracked or Broken Stator Disk)

(c) If no cracked or broken stator disk is found, before further flight, reassemble the brake assembly and, if the piston housing is impression-stamped with the letters "SB," obliterate the existing markings on the piston housing by stamping "XX" over the letters "SB." If paragraph E.(3)(a) or E.(3)(b), as applicable, of Goodrich Service Bulletin 2-1528-32-2 (for airplanes equipped with BFGoodrich brake assembly P/N 2-1528-6); or Goodrich Service Bulletin 2-1530-32-2 (for airplanes equipped with BFGoodrich brake assembly P/N 2-1530-4); both Revision 5; both dated February 19, 2003; as applicable; specifies repetitive inspections, repeat the inspection required by paragraph (b) of this AD at intervals not to exceed those specified in the service bulletin, until paragraph (e) of this AD is accomplished.

Corrective Action (Cracked or Broken Stator Disk)

(d) If any cracked or broken stator disk is found, prior to further flight, replace the brake assembly with a new or serviceable brake assembly; in accordance with Goodrich Service Bulletin 2-1528-32-2 (for airplanes equipped with BFGoodrich brake assembly P/N 2-1528-6); or Goodrich Service Bulletin 2-1530-32-2 (for airplanes

equipped with BFGoodrich brake assembly P/N 2-1530-4); both Revision 5; both dated February 19, 2003; as applicable. If repetitive inspections are required by paragraph (c) of this AD, replacement of all brake assemblies on the airplane with new or serviceable brake assemblies that contain only stator disks stamped with "CHG AI" or "CHG B" or a higher change letter terminates those inspections.

Replacement of Brake Assembly

(e) When the brake assembly has accumulated 700 total landings since its installation or within 50 landings on the airplane after the effective date of this AD, whichever is later, replace the brake assembly with a new or serviceable brake assembly; in accordance with Goodrich Service Bulletin 2-1528-32-2 (for airplanes equipped with BFGoodrich brake assembly P/N 2-1528-6); or Goodrich Service Bulletin 2-1530-32-2 (for airplanes equipped with BFGoodrich brake assembly P/N 2-1530-4); both Revision 5; both dated February 19, 2003; as applicable. If repetitive inspections are required by paragraph (c) of this AD, replacement of all brake assemblies on the airplane with new or serviceable brake assemblies that contain only stator disks stamped with "CHG AI" or "CHG B" or a higher change letter terminates those inspections.

Parts Installation

(f) As of the effective date of this AD, no person may install a BFGoodrich brake assembly on any airplane unless it has been inspected as specified in paragraph (f)(1) or (f)(2) of this AD, and found to be free of cracked or broken stator disks.

(1) For BFGoodrich brake assembly P/N 2-1528-6: Brake assembly must be inspected in accordance with paragraphs (a), (b), and (c) of this AD, as applicable, in accordance with the service information specified in those paragraphs or BFGoodrich Service Bulletin 2-1528-32-3, dated March 23, 2000.

(2) For BFGoodrich brake assembly P/N 2-1530-4: Brake assembly must be inspected in accordance with paragraphs (a), (b), and (c) of this AD, as applicable, in accordance with the service information specified in those paragraphs or BFGoodrich Service Bulletin 2-1530-32-3, dated March 23, 2000.

Alternative Methods of Compliance

(g) In accordance with 14 CFR 39.19, the Manager, Wichita Aircraft Certification Office (ACO), FAA, is authorized to approve alternative methods of compliance for this AD.

Incorporation by Reference

(h) Unless otherwise specified in this AD, the actions shall be done in accordance with the applicable service bulletin listed in Table 1 of this AD.

Table 1.--Service Bulletins Incorporated by Reference

Service bulletin	Revision	Date
BFGoodrich Service Bulletin 2-1528-32-3	Original	March 23, 2000
BFGoodrich Service Bulletin 2-1530-32-3	Original	March 23, 2000
Goodrich Service Bulletin 2-1528-32-2	5	February 19, 2003
Goodrich Service Bulletin 2-1530-32-2	5	February 19, 2003

This incorporation by reference was approved previously by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51 as of June 2, 2004 (69 FR 23093, April 28, 2004). Copies may be obtained from Cessna Aircraft Co., P.O. Box 7706, Wichita, Kansas 67277. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; at the FAA, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Mid-Continent Airport, Wichita, Kansas; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741-6030, or go to:

http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Effective Date

(i) The effective date of this amendment remains June 2, 2004.

▼ Footer Information

Issued in Renton, Washington, on April 16, 2004.
Michael J. Kaszycki,
Acting Manager, Transport Airplane Directorate,
Aircraft Certification Service.
[FR Doc. 04-9380 Filed 4-27-04; 8:45 am]
BILLING CODE 4910-13-P

▼ Comments

CORRECTION: [Federal Register: June 15, 2004 (Volume 69, Number 114); Page 33285-33287; www.access.gpo.gov/su_docs/aces/aces140.html] Go to the attached "pdf" for full correction text. This copy reflects the correction.