



ÚŘAD PRO CIVILNÍ LETECTVÍ  
ČESKÁ REPUBLIKA  
**Sekce technická**  
letišťe Ruzyně, 160 08 Praha 6  
tel: 233320922, fax: 220562270

## PŘÍKAZ K ZACHOVÁNÍ LETOVÉ ZPŮSOBILOSTI

**Číslo: 2008-0069**

Datum účinnosti: 25. dubna 2008

**Saab AB, Saab Aerosystems**  
SAAB SF340A, 340B

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Tento PZZ je vydáván pro výrobek transferovaný pod působnost EASA


Na základě rozhodnutí EASA je následující Příkaz k zachování letové způsobilosti závazný pro všechny výrobky provozované v EU, na které se daný PZZ vztahuje.

Provedení PZZ, který se vztahuje podle typu a výrobního čísla na výrobek je pro provozovatele/vlastníka letadla zapsaného do leteckého rejstříku závazné. Neprovedením PZZ ve stanoveném termínu dojde ke ztrátě letové způsobilosti výrobku.

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*Poznámky:*

- Provedení tohoto PZZ musí být zapsáno do provozní dokumentace letadla.
- Případné dotazy týkající se tohoto PZZ adresujte na ÚCL sekce technická.
- Pokud to vyžaduje povaha tohoto PZZ, musí být zapracován do příslušné části dokumentace pro obsluhu, údržbu a opravy letadla.

<b>EASA</b>	<b>AIRWORTHINESS DIRECTIVE</b>	
	<p><b>AD No : 2008-0069</b></p> <p><b>Date: 11 April 2008</b></p>	
No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise agreed with the Authority of the State of Registry.		
<b>Type Approval Holder's Name :</b> Saab AB, Saab Aerosystems	<b>Type/Model designation(s) :</b> SAAB SF340A and 340B	
TCDS Number : EASA A.068		
Foreign AD : Not applicable		
Supersedure : None		
<b>ATA 71</b>	<b>Power Plant – Forward Engine Cowl Door – Inspection / Reporting</b>	
<b>Manufacturer(s):</b>	Saab AB, Saab Aerosystems	
<b>Applicability:</b>	SAAB SF340A Serial No's 004 thru 159 and SAAB 340B Serial No's 160 thru 459	
<b>Reason:</b>	<p>Field experiences have revealed cracks in the frames and closing angle on the forward engine cowl door NS STA 203 and 250.</p> <p>In case of a damaged frame and/or closing angle, the forward engine cowl door can loosen during flight and depart from the aircraft.</p> <p>This AD is issued to require a detailed inspection to find out if there are any cracks in the frames and/or the closing angles. The inspection is on four points on each of the forward engine cowl doors.</p>	
<b>Effective Date:</b>	25 April 2008	
<b>Compliance:</b>	<p>1. Required as indicated, unless accomplished previously: Within 1 000 flight hours from the effective date of this Airworthiness Directive, perform inspection in accordance with SB-340-71-060.</p> <p>2. When during inspection, cracks, deformations or damage are found, before next flight, repair according to Saab 340 Structure Repair Manual (SRM) or consult Saab for corrective action.</p> <p>3. Within 10 days, report the inspection results to Saab AB, Saab Aerosystems.</p>	

Ref. Publications:	SAAB Service Bulletin SB-340-71-060, dated 08 February 2008. The use of later approved revision of this document is acceptable for compliance with the requirements of this AD.
Remarks :	<ol style="list-style-type: none"><li>1. If requested and appropriately substantiated, EASA can accept Alternative Method of Compliance for this AD.</li><li>2. This AD was posted on 20 March 2008 as PAD 08-036 for consultation until 04 April 2008. No comments were received during this period.</li><li>3. Enquiries regarding this Airworthiness Directive should be referred to the Airworthiness Directive Focal Point, Certification Directorate, EASA. E-mail <a href="mailto:ADs@easa.europa.eu">ADs@easa.europa.eu</a></li><li>4. For any question concerning the technical content of the requirements in this AD, please contact Saab at Telephone: +46 13 185591; Fax: +46 13 184874 or E-mail: <a href="mailto:technical.support@sat.saabgroup.com">technical.support@sat.saabgroup.com</a></li></ol>