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| EASA | AIRWORTHINESS DIRECTIVE | |
|  | <p>AD No : 2007-0101</p> <p>Date: 13 April 2007</p> | |
| <p>No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise agreed with the Authority of the State of Registry.</p> | | |
| <p>Type Approval Holder's Name : EADS SOCATA</p> | <p>Type/Model designation(s): TB 9, TB 10, TB 20 and TB 21</p> | |
| <p>TCDS Number: DGCA France No 165</p> | | |
| <p>Foreign AD Nr: Not applicable</p> | | |
| <p>Supersedure: DGAC F-1991-143 R1</p> | | |
| <p>ATA 52</p> | <p>Doors – Cabin Access Door Hooks – Inspection / Replacement</p> | |
| <p>Manufacturer:</p> | <p>SOCATA, EADS SOCATA</p> | |
| <p>Applicability:</p> | <p>TB 9, TB 10, TB 20 and TB 21 aircraft, all serial numbers equipped with aluminium alloy made cabin access door hooks.</p> | |
| <p>Reason:</p> | <p>The superseded DGAC France F-1991-143 R1 Airworthiness Directive was issued to address the development of cracks in aluminium-alloy made cabin access door hooks that if not corrected could lead to a cabin door opening in flight.</p> <p>The aim of this new AD is to reduce the applicability and modify the compliance actions of the superseded DGAC France F-1991-143 R1 AD, excluding aircraft equipped with steel-made hooks which bring a definitive solution to correct the unsafe condition addressed in the present AD.</p> | |
| <p>Effective Date:</p> | <p>27 April 2007</p> | |
| <p>Compliance:</p> | <ol style="list-style-type: none"> 1. On the next scheduled 100 hour inspection or annual inspection, whichever occurs first, Check the aluminium-alloy made cabin access door hooks for cracks as specified in the accomplishment instructions of the EADS SOCATA Service Bulletin No 10-058. 2. Repeat thereafter inspections of § 1. at intervals not exceeding 110 flight hours. | |

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| | <p>3. Discard the AU4G aluminium-alloy made cabin access door hooks which have accumulated more than 1500 hours of Time in Service and at the latest on 31st December 2011, remove any AU4G hooks and replace them by steel ones, whose part number is P/N TB 10 250 72 101.</p> <p>Note: Replacing AU4G aluminium-alloy made cabin access door hooks by steel ones is considered as a terminating action for this AD's requirements.</p> |
| <p>Ref. Publications:</p> | <p>EADS SOCATA Service Bulletin No 10-058 Amdt 2.</p> <p>Any further approved revision of this document is acceptable.</p> |
| <p>Remarks:</p> | <p>1. If requested and appropriately substantiated the responsible EASA manager for the related product has the authority to accept Alternative Methods of Compliance (AMOC) for this AD.</p> <p>2. Required actions and the risk allowance have granted publication and notification of an immediate AD, ruling out the public consultation process.</p> <p>3. Enquiries regarding this AD should be addressed to Mr. M. Capaccio, AD Focal Point, Certification Directorate, EASA. E-mail: ADs@easa.europa.eu</p> <p>4. For any questions concerning the technical content of the requirements in this AD, please contact:</p> <p>EADS SOCATA - Direction des Services - 65921 Tarbes Cedex 9 - France Tel. : +33 (0)5 62 41 73 00 Fax : +33 (0)5 62 41 76 54 Or for the US of A : SOCATA AIRCRAFT, INC. - North Perry Airport - 7501 South Airport Rd. Pembroke Pines, FL 33023 - United States of America Tel. : 1 (954) 893 1400 Fax : 1 (954) 964 4141</p> |